

Smart Structures and Systems

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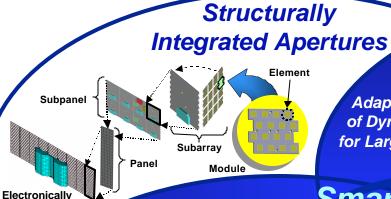
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Smart Structures Focus Areas



Adaptive Compensation of Dynamic Deformations for Large Antenna Systems



Adaptive Structures

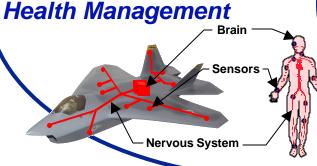
- Vibration and Load Alleviation
- Shape Change
- Acoustic Suppression

Load Bearing
Structures with
Integrated Electronic
and Photonic Systems

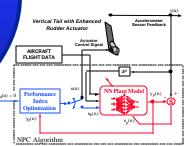
Smart Structures with Integrated, Multi-Functional Capability

Adaptive Active Structural Control

Structural Health Manageme



Advanced Information
Processing for Structural
Health Monitoring



Intelligent Systems

- Neural Networks for Control and
- Information Processing



Steered Array (ESA)

Smart Structures & Systems



Technology Transition Strategy



	Technology Readiness Level (TRL)					
2004-7	System Implementation	System Test, Flight and Operations	9	Actual system "Flight Proven" through successful mission operations		
2002	p		8	Actual system completed and "Flight Qualified" through test and demonstration		
	Flight and At-Sea Demonstrations	System/Subsystem Development	7	System prototype demonstrated in operational environment		
2000	CARADCORI	Technology Demonstration	6	System/subsystem model or prototype		
1997	SAMPSON	Tashualami	5	Component and/or breadboard validation in a relevant environment		
		Technology Development	4	Component and/or breadboard validation in a laboratory environment		
	SPICES	Research to Prove Feasibility	3	Analytical and experimental critical function and/or characteristic proof-of-concept		
		Basic Technology Research	2	Technology concept and/or application formulated		
1993			1	Basic principles observed and reported		







Synthesis and Processing of Intelligent Cost Effective Structures



Develop Cost Effective Material **Processing and Synthesis** Technologies Which Will Enable New Products Requiring Active Vibration Suppression and Control Systems to be Brought to Market



Sensor Development

- Fiber Optics
- Piezo sensors
- Microaccelerometers

<u> Active Mount Concept</u>

1-4 KHz Vibration Control (Commercial Application) 5-100 Hz Vibration Control (Military Application)

Hierarchic Control Systems

- Local Damping Augmentation
 - Global Vibration Control
 - Electrical Shunting
 - Frequency Shifting

Manufacturing/Integration

- Embedding Techniques
- Automated Fabrication
- Advanced Nervous Systems

Heterogeneous Modeling

- Superelement Techniques
- Interfaces to Control Design
 - Nonlinear Capabilities

- **Actuator Development**
- High Force PZT Systems
- New forms of Shape Memory Alloys
 - Survivable Subsystems

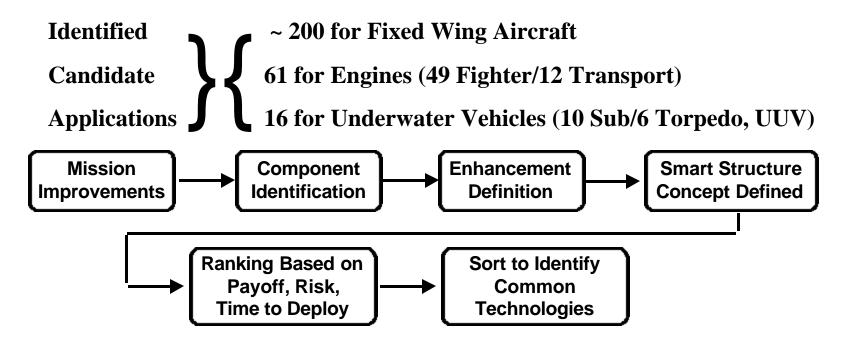


Final Fabrication and Demonstrations

- Demonstrate 30dB attenuation
 - Quantification of Best Practices and Cost Effective Procedures



SPICES II Applications Screening Process



Concepts Were Quantitatively Screened and Down-selected to:

- 3 Fixed Wing Aircraft
- 4 Marine
- 4 Gas Turbine Engine (2 Tactical, 2 Transport)

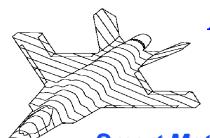


Downselected Aircraft Smart Materials Concepts



Downselected Concepts	High Performance Platforms	Details	Core Technologies
Control Surface Shape Change - Quasi-static	Supersonic Covert Penetrator, Unmanned Tactical Aircraft, Missiles, Munitons	Low Rate, Smooth Surface Deflection of LE & TE Flaps	SMA & Flex Skin
Inlet Shape Change	Light Weight Strike Fighter, Supersonic Covert Penetrator	Capture Area Control, Ramp Angle Change, Lip Blunting	SMA & Flex Skin
Door and Control Surface Gap Filling	Supersonic Covert Penetrator, Unmanned Tactical Aircraft	Close Gap Between Deflected Flap and Wing Trailing Edge	Flex Skin
Wing Lift Increase	Military Aircraft, Fighter and Transport,e.g., Blended Wing-Body	Replace Leading Edge Slats with Shape Change	SMA & Flex Skin
Control Surface Shape Change - High Rate	Unmanned Tactical Aircraft	Actual or Virtual Shaping of Flaps, Ailerons, Rudders	Inchworms & Flex Skin, Synthetic Jets
Inlet Approach Surface Boundary Layer Control	Blended Wing-Body	Energize Boundary Layer over Wing/Fuselage Surface	Synthetic Jets
Control Surface Boundary Layer Management	Supersonic Covert Penetrator	Energize Boundary Layer over Deflected Flaps and Ailerons	Synthetic Jets
Inlet Diffuser Boundary Layer Management	Fighter Aircraft, Supersonic Covert Penetrator	Energize Boundary Layer Inside High Offset Inlet Duct	Synthetic Jets
Nozzle Fluidics Thrust Vectoring	Supersonic Covert Penetrator	Achieve High-Rate Jet Flow Turning and Area Control	Synthetic Jets
Wing & Tail Shaping for Maneuvering Enhancement	Advanced Cruise Missile, Advanced Munitions	High-Rate Surface Shape Change for Maneuvering	Piezo Sheets, Stacks, Inchworms
Moldline Control in Maneuvering Flight	Unmanned Tactical Aircraft	Maneuverability Without Deflecting Control Surfaces	Synthetic Jets
Weapons Bay Noise/Wake Control	Supersonic Covert Penetrator, Strategic Supersonic Bomber	Active Cancellation of Cavity Acoustics	Synthetic Jets
Wing Drag Reduction	Supersonic Covert Penetrator	Adjustable Wing Camber	Synthetic Jets
Nozzle Area Control	Supersonic Covert Penetrator	Control Jet Area Inside Fixed- Aperture Nozzle	High Temperature SMA





New Mission Is Enabled by Variable Geometry Smart Structures Inlet



Smart Materials Actuated Variable Geometry Inlet Provides:

Compression Ramp
Capture Area Control
Lip Blunting

Subsonic Geometry

Supersonic Geometry



Combat at Dash Alt
•1 Min @ Max A/B

Subsonic Cruise
• Best Mach/Alt

Supersonic Dash

Variable Geometry Inlet Provides
20%+ Increase in Mission Radius
(relative to fixed inlet baseline design)

Loiter

• 20 min
• Sea Level

Enables Strike Aircraft Optimized for Subsonic Interdiction Mission to Also Perform Supersonic Intercept Mission







SAMPSON Modified F-15 Inlet with Smart Structures Actuation



Actuator

CMT

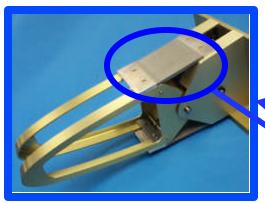
elastomer

not shown

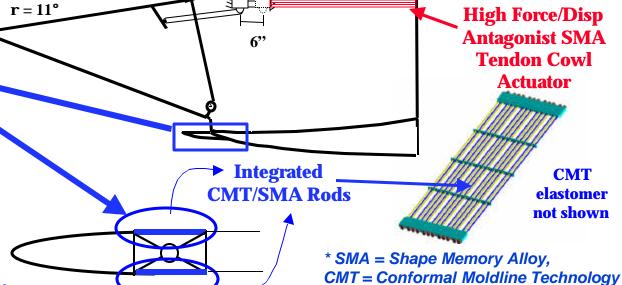


- First <u>full-scale</u> demonstration of high force & displacement smart materials actuation.
- First integration of SMA* rod actuators within compliant structure configurations.
- First applications demo of Pd doped SMA (high transition temperature)

First entry in Langley 16-Ft Transonic April 2000



Adaptive inlet provides >20% increase in mission radius re: fixed geometry inlet (F-16, F/A-18)









Projects Demonstration

SAMPSON = Smart Aircraft and Marine



Technology Readiness Levels

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Technology Demonstration	6 System/subsystem model or prototype demonstration in a relevant environment
Technology	5 Component and/or breadboard validation in a relevant environment
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Research to Prove Feasibility	3 Analytical and experimental critical function and/or proof-of-concept
Basic Technology Research	2 Technology concept and/or application formulated
100001011	1 Basic principles observed and reported

Low rate, modest shape change

Primary flight control

Massive config. shape change







Technology Readiness Levels Required to be Ready for Flight Test

System Test, Flight and Operations		Actual system "Flight Proven" through successful mission operations
·	8	Actual system completed and "Flight Qualified" through test and demonstration
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Research to Prove Feasibility Basic Technology Research		Analytical and experimental critical function and/or proof-of-concept
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TRL 7 min. for SOF critical appl. ready for full scale flight test

TRL 6 min. for SOF critical appl. ready for sub-scale flight test, or Non-SOF critical applications ready for full scale flight test

TRL 5 min. for Non-SOF critical applications ready for sub-scale flight test

SOF = Safety of Flight









Acceptable Technology Risk Level Varies with Program Maturity

Program Development Phase

TRL	Readiness Level Completed	Concept Exploration & Definition	Demonstration/ Validation	Engineering / Manufacturing Development	Production/ Deployment	Operations/ Support
9	Production Flight Proven					
8	Flight Test Qualified					
7	Prototype Test in Operational Env					
6	System Test in Relevant Env					
5	Component Test in Relevant Env	Low Risk				
4	Component Test in Lab Env	Medium Risk				
3	Proof of Concept Testing	High Risk	Unacceptable Risk			
2	Concept/Application Formulated					
1	Basic Principles Reported					

Typical Program Lower Limit on Risk

May be Acceptable to Solve a Problem

May Not Be Actively Tracked Under Risk Management







Transition Conclusions

- Smart structures transition requires high TRL if transition is during or after E&MD
 - O Risk (real and perceived)
 - O Impact to re-qualification
 - Impact to operations and support

All conspire to overwhelm benefits and kill the business case (unless it solves a problem)

- □ Need to enable paradigm shift, but MUST have a viable mission
 - O Entirely new mission
 - O Multi-mission replace two or more systems
- □ Suggested approach
 - O Clean sheet design
 - O UAV to reduce acceptable risk and program cost
 - Phased approach to provide intermediate, nearer term capability
 (low rate, modest shape change first, then massive shape change and primary flight control)

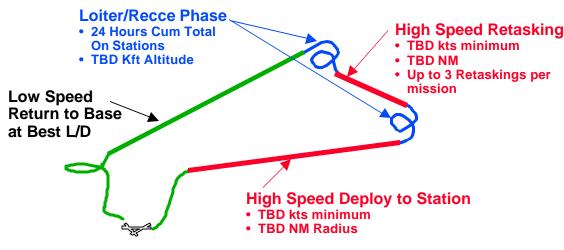






Smart Structures UAV Mission

- □ Long duration UAV with high speed capability
 - O ISR capabilities E/O, SAR / large ESA
 - High Speed for Deploy / Retasking (>M=0.7, >M=1?)
 - O Attack capability on Mobile / Relocatable Targets
 - O High Altitude (trades for survivability)
- □ Wing shape change span, AR, sweep, area
 - O Low Rate shape change is nearer term enabler
 - O Massive shape change should be product improvement
 - O Primary flight control benefits are secondary
- □ Demo on no smaller than half scale





Global Hawk ISR Capability





